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EVALUATION OF EFFECT CHANGING TEMPERATURE ON LAMB-WAVE BASED STRUCTURAL HEALTH MONITORING

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Abstract: The purpose of this paper is to investigate the effect of temperature change on the Lamb wave-based SHM method. This study evaluates the Lamb wave method's ability to detect damage to an AL2024-T3 sheet, assessed by a near-surface hole. Lamb waves are created via numerical simulation with the commercial Finite Element (FE) package ABAQUS. In this study, the Lamb wave-based SHM method using displacement responses is used. The results indicate that this method is able to detect a near-surface hole in the AL2024-T3 sheet as well as its location, with close approximation. Subsequently, the AL1100 sheet was investigated for changes in temperature from this method, which was evaluated over a temperature range of -200°C to 204°C . The results show that temperature change in the range of -200°C to 93°C has no effect on the displacement responses. However, the graphs related to temperature change of more than 149°C do not overlap with the reference temperature. Hence, it has been concluded that Lamb waves can be used as an SHM method in the temperature range of -200°C to 93°C without having to worry about the effects of temperature change on the results.

Keywords: structural health monitoring; lamb wave, temperature; AL2024-T3 sheet; AL1100 sheet.

1. INTRODUCTION

The safety is extremely important in aviation industry. Any defect in aircraft structures is not acceptable due to the possibility of loss of life and capital involved. Conditions sustaining such as; high loads, fatigue cycles and extreme temperature differentials can cause defect in aircraft structures. Inspection of aircraft, play a very important role in aviation safety [1-3]. Non-destructive testing (NDT) methods are used to ensure the safe operation of aircraft structures. Ultrasonic Testing (UT) is one of commonly used methods for NDT. UT is used to detect the presence of defects of the materials and for analyze and characterize some important properties of materials such as microstructure, mechanical properties of materials [4-6]. Lamb waves are ultrasonic waves in thin plates [7]. Lamb waves are guided acoustic waves which propagate in the plane of a plate and, like bulk acoustic waves, their interaction

with defects means that they can be used for inspection purposes [8]. In recent years, lamb wave based damage detection has been widely investigated. Larger area scanning, low attenuation, easy implementation using piezoelectric (PZT) transducers, online Structural Health Monitoring (SHM) are some of several advantages of lamb wave based technique [9, 10].

In recent years, the use of simulation models to develop ultrasonic based damage detection techniques has become very popular, because it's a less expensive and time-consuming procedure than real structures or experimental models [11, 12]. Also, experimental setup is also a fairly difficult process. Since the Finite Element Method (FEM) has been widely accepted as an analysis tool in SHM the above mentioned constraints can be overcome by using a validated FE model to simulate the real structure [13]. The FEM is the best method when complex damage, geometry or boundary is involved. A transient dynamic FE simulation of lamb wave with piezoelectric transducers for damage detection in a composite plate

is carried out by Wenzhong and Li [14] and they demonstrated that the numerical simulation is the effectiveness of the approach [14]. Nieuwenhuis et al [15], presented two-dimensional FEM transient simulations of the source region, we show how those results can guide a designer in choosing transducer dimensions for mode selectivity [15]. In this study, a transient dynamic FE simulation of lamb wave for damage detection in 2024-T3 aluminum sheet is carried out. The FE analysis was conducted using the commercial FE package ABAQUS. In this study, the Lamb wave-based SHM method using displacement responses is used. To use lamb wave based SHM, it's necessary to consider the effect of environmental factors such as changing temperature on the lamb waves. Several researches on changing temperature on UT technique have been done. Most of previous researches have been experimentally [4, 16]. One of the earliest works on the temperature effects on Lamb wave signals was performed by Francesco Lanza di Scalea and Salvatore Salamone [17]. The model is used to predict the S0 and A0 response spectra in aluminum plates for the temperature range of -40 to $+60^{\circ}\text{C}$. The study shows substantial changes in lamb wave amplitude response caused solely by temperature excursions [17]. A pitch-catch arrangement between 50 and 150 kHz on sandwich panels at low temperatures (-90°C) was performed by Blaise and Chang [18]. The authors observed a decrease in wave amplitude and time of flight from ambient temperature to -90°C , and they developed an empirical model to fit the experimental data [13]. Konstantinidis et al. [19] studied the variation in wave amplitude between 22 and 32°C for various portions of lamb wave measurements in aluminum plates and showed that temperature changes result in shifts in wave arrival times and in center frequency of the received waves under constant excitation frequency. Lu and Michaels (2005) and Michaels and Michaels (2005) studied the difference in various damage-sensitive features of diffuse lamb wave signals at temperatures varying between 5 and 40°C , and showed that a changing time of flight, which was attributed to thermal expansion and Young's modulus of the substrate. [20, 21] Temperature compensation of guided wave based supervised SHM systems is compulsory for components exposed to environmental conditions by DAN et al. [22]. They showed that the wave variables are transferred from the wave coefficients, the shear and longitudinal velocities [22]. Han et al. [23] studied FE Analysis of lamb wave propagation in a thin aluminum plate. Simulations of isothermal tests are conducted over a temperature range of $0-190^{\circ}\text{F}$ using 100 and 300 kHz as excitation frequencies. They showed that elevated temperatures delay the Lamb wave propagation, although the delays are found to be minimal at the temperatures tested [23]. Dhutti et al.

[24], studied application of Ultrasonic Guided Wave (UGW) technique for in-service SHM of high temperature (HT) pipework. They settled the system on a pipe operating at up to 200°C for over 1 year [24].

The purpose of the study is to evaluate effect of changing temperature on SHM using lamb waves. For this propose, sensitivity analysis of the lamb wave to temperature changes range of -200°C to 204°C in AL 1100 sheet is evaluated. Assuming that the location of inspection is located at the hottest and coldest places on Earth, and that we want to inspect a component at these places, then -70°C to $+70^{\circ}\text{C}$ change in the temperature range is enough to investigate the effect of changes in the temperature on the lamb wave. However, in this paper, a greater temperature range was selected to achieve wider and more accurate judgments about the effect of changes in the temperature on the lamb wave. In this study, to evaluate the ability of lamb waves based SHM method and effect of changing temperature on SHM using lamb waves; two different types of aluminum (1100 and 2024-T3) are used. SHM to assess the structural condition plays a vital role for safe and efficient operation of aircraft. 2024-T3 aluminum sheet is widely used in the aviation industry. Thus, in this study, to evaluate the ability of the lamb wave-based SHM method, a 2024-T3 aluminum sheet is selected. To evaluate the sensitivity analysis of the lamb waves to temperature changes, an AL 1100 sheet is selected, because it has a high coefficient of thermal expansion. The remainder of the paper is organized into four sections: In Section II, effect change temperature on mechanical properties will be discussed. Numerical simulation is presented in Section III. The conclusion is reported in Section IV.

2. THEORY EFFECT OF CHANGING TEMPERATURE ON LAMB WAVE

Particularly, solids experience upon change of temperature, modifications in their mechanical properties, of interest being here the elastic and shear moduli and the density, which, at their turn cause changes in the two main acoustic properties of any given material, acoustical absorption and speed of sound [25].

2.1. Wave propagation

Andrews 2007, indicate that plate waves are result from the conversion between transverse (T) and longitudinal (L) modes. Wave propagation is dependent on the density (ρ) and elastic properties of a medium. The longitudinal wave speed is characterized by the Young's modulus (E) as [18]:

$$c_L = \sqrt{\frac{E}{\rho}}. \quad (1)$$

Similarly, the transverse shear wave speed is characterized by the shear modulus (G) as [18]:

$$c_T = \sqrt{G/\rho}. \tag{2}$$

2.2. Effect of changing temperature on mechanical properties of sample

Xu and Zhishen showed that the changes caused by temperature were analyzed based on the following aspects: (1) change of the passion rate of the aluminum alloy; (2) changes of the elastic modulus of the aluminum alloy. Scalea and Salamone indicate that a linear dependence of each property on temperature is assumed as following equation (3) [26]:

$$P(T) = P(T_0) + \frac{\partial P(T)}{\partial T} \Delta T, \tag{3}$$

where P represents one of the mechanical properties of sample, such as Young’s modulus E , Poisson’s Ratio ν , shear modulus G , and bulk modulus K , T is the generic temperature, T_0 is the ambient temperature 20°C, and $\frac{\partial P(T)}{\partial T}$ is the sensitivity to temperature [19]. Also, Nowacki (1962) has demonstrated that the temperature dependence of the modulus of elasticity for most of engineering materials is given [27]:

$$E(T) = E_0(1 - \gamma T), \tag{4}$$

where E_0 is the value of Young’s modulus at the reference temperature, i.e. $T = 0$, and γ is the slope of the variation of E with T [27].

3. DAMAGE DETECTION USING LAMB WAVE

In this section, which first evaluates the ability of the damage detection of lamb waves, a 2024-T3 aluminum alloy sheet with a lamination defect is assessed. Aluminum alloys are widely used for aerospace applications including primary aircraft structures because of lightweight or corrosion resistance is required. The 2024-T3 aluminum sheet is one of the best known high-strength aluminum alloys. It is thought of as the aircraft alloy because of its strength. It has excellent fatigue resistance. Typical uses for 2024-T3 aluminum sheet are aircraft skins, cowls, aircraft structures. Alloy 2024-T3 sheet products are used extensively in commercial and military aircraft for fuselage skins, pressure cabin skins, wing skins and engine areas where elevated temperatures to 250°F (121°C) are often encountered [28]. Then the sensitivity analysis of the lamb waves to temperature changes in an AL 1100 sheet is evaluated. Lamb waves are created via numerical simulation with the commercial FE package ABAQUS. The displacement responses of nodes are obtained.

3.1. Simulation modelling and analysis of intact structure

First, a FE model of the 2024-T3 aluminum alloy sheet is created using ABAQUS. A 2024-T3 aluminum alloy sheet with a near-surface defect (a hole) is shown in Figure 1. The material properties are listed in Table 1. In an experimental test, data are obtained from sender and receiver sensors. In a simulation test, two nodes are determined on the surface of the plate as sender and receiver sensors. The locations of the sender and receiver sensors are shown in Figure 2. Multiple runs show that the type of mesh has a non-significant effect on the results, while free meshing is more suitable on the damaged section. Thus, free meshing with triangular and tetrahedral elements is utilized (Figure 3). Then lamb waves are created and the displacement responses of the receiver node are obtained for an intact structure (Figure 4). The result of using the lamb wave for an intact structure is shown in Figure 5.

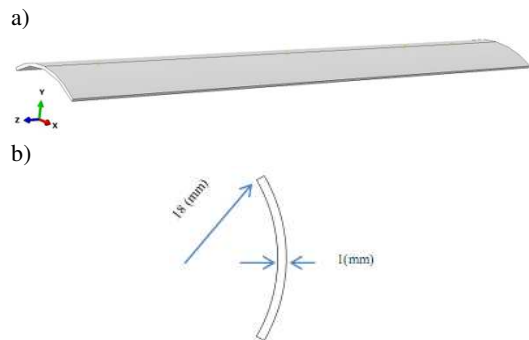


Fig. 1. a) & b) A FE model of beam was created using ABAQUS.4

Tab. 1. Material properties for 2024-T3 aluminium alloy

| Material | 2024-T3 aluminium alloy |
|---------------------------------|-------------------------|
| Density (ρ).(g/cc) | 2.7 |
| Poisson's Ratio (ν) | 0.33 |
| Modulus of Elasticity (E) (Gpa) | 70 |

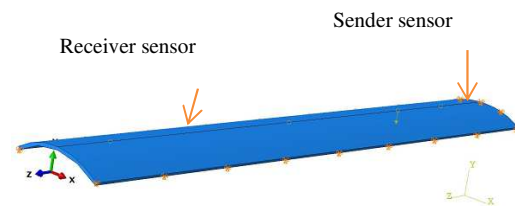


Fig. 2. The location of sender and receiver sensors

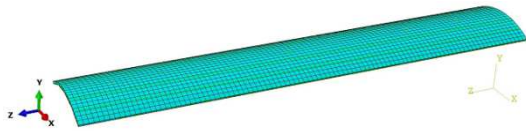


Fig. 3. Free meshing technique with linear tetrahedral and hexahedral elements

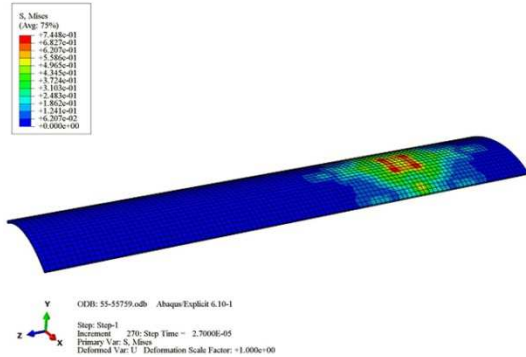


Fig. 4. Lamb wave are created in Al 2024-T3 sheet in ABAQUS

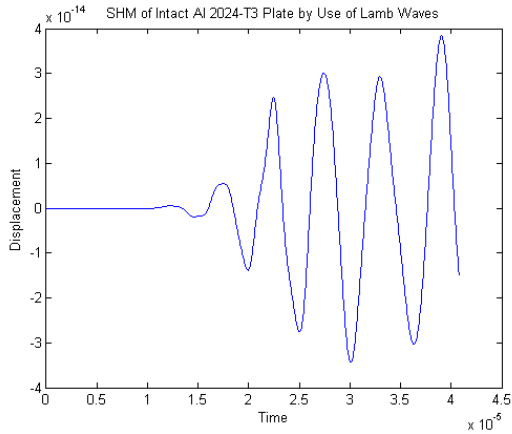


Fig. 5. SHM of intact AL2024-T3 sheet using lamb wave

3.2. Simulation modelling and analysis of damaged structure

First, a hole defect is created in the near-surface of the model. The type of defect is a hole with 0.1 mm diameter. The size of the defect is shown in Figure 6. The elements of type, the total number of elements and nodes for the undamaged model and damage scenarios that were created in the model are shown in Table 2. Then, lamb waves are created and the displacement responses of receiver node are obtained for the damaged structure. The result of using the lamb wave for the intact structure is shown in Figure 7.

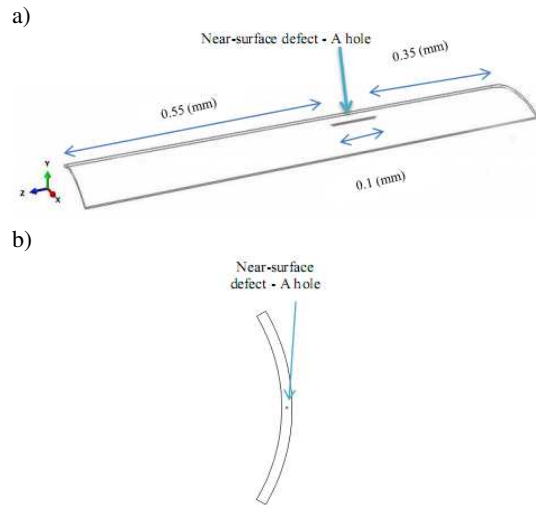


Fig. 6. A defect was created in model

Tab. 2. The elements of type, the total number of elements and nodes for undamaged model and damage scenario that was

| Structure Status | Undamaged Model | Damaged Model |
|-----------------------|--|--|
| Total number of nodes | 4242 | 15351 |
| Total number elements | 2000 | 46638 |
| The elements of type | linear hexahedral elements of type C3D8R | linear tetrahedral elements of type C3D4 |

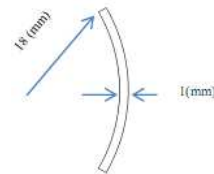


Fig. 7. A FE model of beam was created using ABAQUS.4

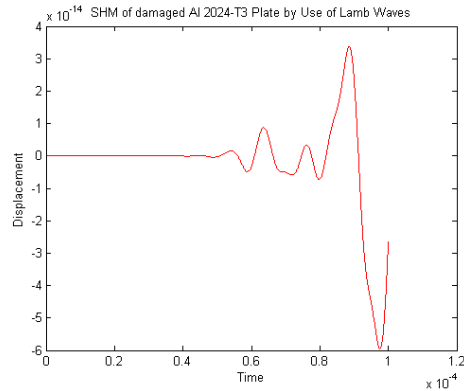


Fig. 8. SHM of damaged AL2024-T3 sheet using lamb wave

3.3. Result and discussions of damage detection using the lamb wave

As shown in Figures 5 and 7, the shape of the wave propagation process of the intact structure is different from the damaged structure. The wave propagation process of the intact structure is more regular. With the damaged structure, the wave propagation process is more irregular than with the intact structure. Also, the lamb wave in the intact structure is more powerful than in the damaged structure (Figure 8). These differences will lead to identifying damage. As seen in Figure 8, the lamb wave is able to detect a near-surface hole in an AL2024-T3 sheet.

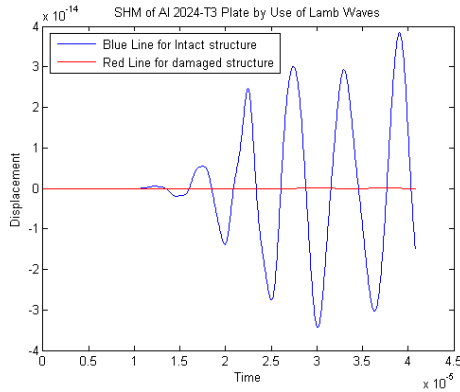


Fig. 9. Compare SHM for intact and damaged AL2024-T3 sheet using lamb wave

3.4. Effect of the temperature on the lamb wave

In this section, sensitivity analysis of the lamb wave to temperature changes in AL 1100 sheet is evaluated. Aluminum 1100 is one of the most widely used alloys, with excellent forming properties in annealed states and may be suitable for bending, spinning, drawing, stamping, roll forming, and many other applications. Typical applications include chemical storage, processing equipment, kitchen utensils, and general sheet metal work. The material properties are listed in Table 3. As shown in Figure 9, to evaluate the effect of changing temperature on lamb waves, the results of analysis of an intact structure obtained by the lamb wave method at a temperature of 21°C are selected as the reference temperature and initial situation ($D_{LW,j}^i$) for the inspection process. Then, the intact structure is evaluated in a temperature range of -200°C to 204°C. The analysis results of an intact structure obtained by the Lamb wave method in a temperature range of -200°C to 204°C are selected as the secondary status ($D_{LW,j}^s$) for the inspection process. The lamb wave method is not sensitive to temperature change, if the results should have overlap; otherwise, it is sensitive to temperature change. The displacement responses of intact structure are obtained via numerical simulation with the commercial FE package ABAQUS. The

results of the investigation into the Lamb wave under temperature changes between -200°C and 203°C is shown in Figure 10. The figure shows that the graphs are almost overlapping for the range of -200°C to 93°C. In other words, temperature changes in the range of -200°C to 93°C had no effect on the response of displacement. However, the graphs relating to temperature changes of more than 149°C do not overlap with the reference temperature. It was been found that lamb waves can be used as an SHM method in the temperature range of -200°C to 93°C, without having to worry about the effects of temperature changes on our results.

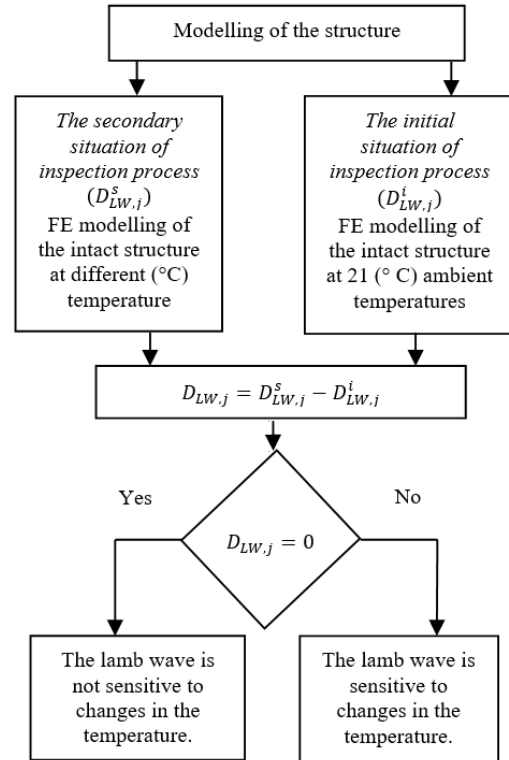


Fig. 10. Flowchart of the method used to investigate temperature changes effect on the lamb wave method

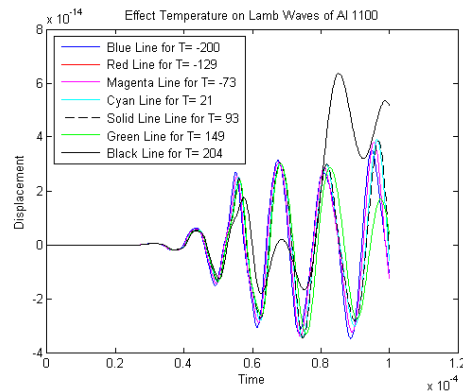


Fig. 11. Effect changing temperature on the lamb wave under changes temperature range of -200°C to 204°C

Tab. 3. Material properties for aluminium 1100

| Material | Aluminium 1100 |
|-------------------------------------|----------------|
| Density (ρ).(g/cc) | 2.71 |
| Poisson's Ratio (ν) | 0.33 |
| Modulus of Elasticity (E) (Gpa) | 70 |

4. CONCLUSIONS

In this paper, effects of temperature on SHM using lamb waves in AL2024-T3 plate were investigated. In this study, the evaluation of effect changing temperature on lamb-wave based SHM using the displacement responses of nodes is used. This study evaluates the Lamb waves method's ability to detect damage to an AL2024-T3 sheet, assessed by a near-surface hole. Lamb waves were created via numerical simulation with the commercial FE package ABAQUS. The results indicate that the IPV method is able to detect a near-surface hole in the AL2024-T3 sheet as well as its location, with close approximation. An AL 1100 sheet was investigated for the effect of changes in temperature on the lamb waves based SHM method, evaluated over a temperature range of -200°C to 204°C . The results show that temperature changes don't effect on displacement responses, thus neither are SHM analysis results affected by temperature changes. In other word, the Lamb waves are not sensitive to changes in temperature range of -200°C to 204°C . This study highlighted the importance of applying simulation methods to develop Non-Destructive Testing (NDT) techniques, especially for evaluation the effect changes in environmental temperature.

Nomenclature

Symbols

| | |
|------------------------------------|------------------------------|
| E | – Young's modulus |
| K | – bulk modulus |
| P | – mechanical properties |
| T | – generic temperature |
| c_L | – longitudinal wave |
| c_T | – transverse shear |
| T_0 | – ambient temperature |
| ν | – Poisson's Ratio |
| $\frac{\partial P(T)}{\partial T}$ | – sensitivity to temperature |

Greek letters

| | |
|----------|--------------------------|
| ρ | – density |
| γ | – slope of the variation |

Acronyms

| | |
|-----|--------------------------------|
| HT | – High Temperature |
| UT | – Ultrasonic Testing |
| FEM | – Finite Element Method |
| NDT | – Non-destructive testing |
| SHM | – Structural Health Monitoring |
| UGW | – Ultrasonic Guided Wave |

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Biographical note



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